Civilian Spatial Disorientation Mishap

Experience



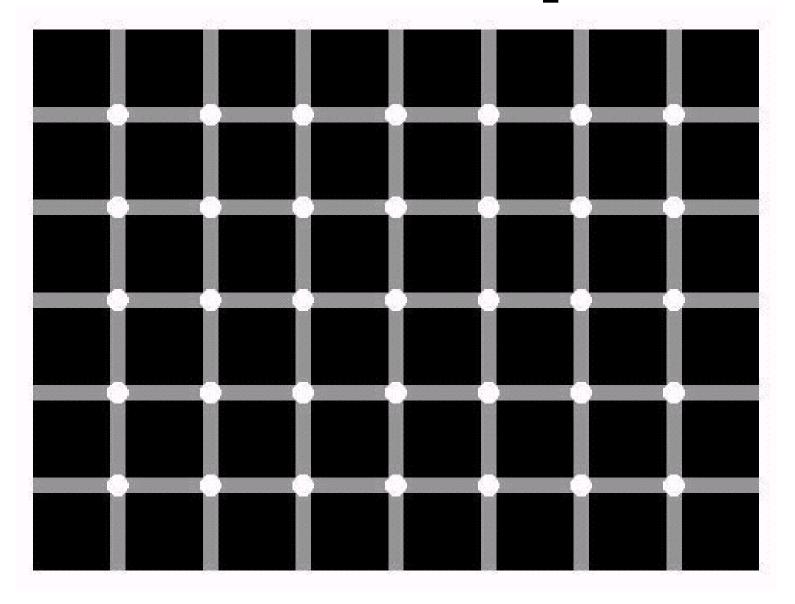
ephen J.H. Véronneau, M

Research Medical Officer

Team Coordinator, Aircraft Accident Research

FAA Civil Aeromedical Institute

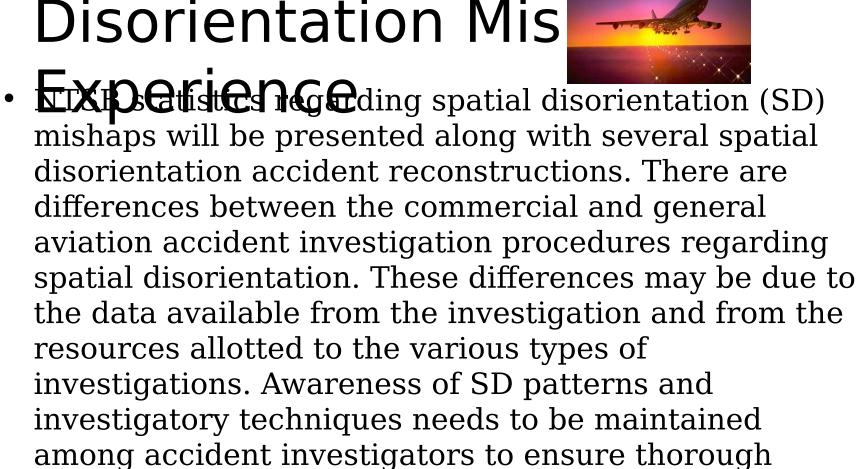
Find the Black Spot



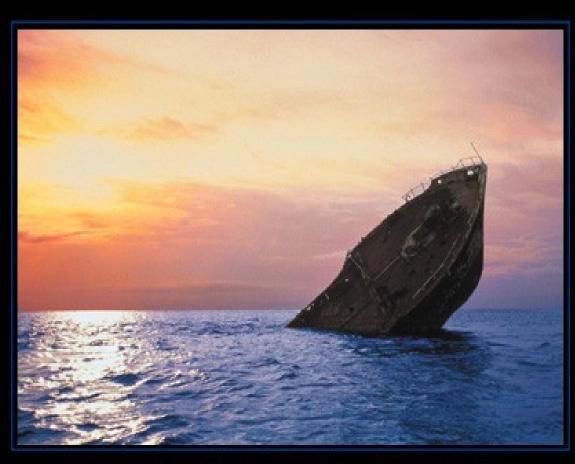
Civilian Spatial Disorientation Mis

epidemiology of SD mishaps.

investigations and accurate information regarding the



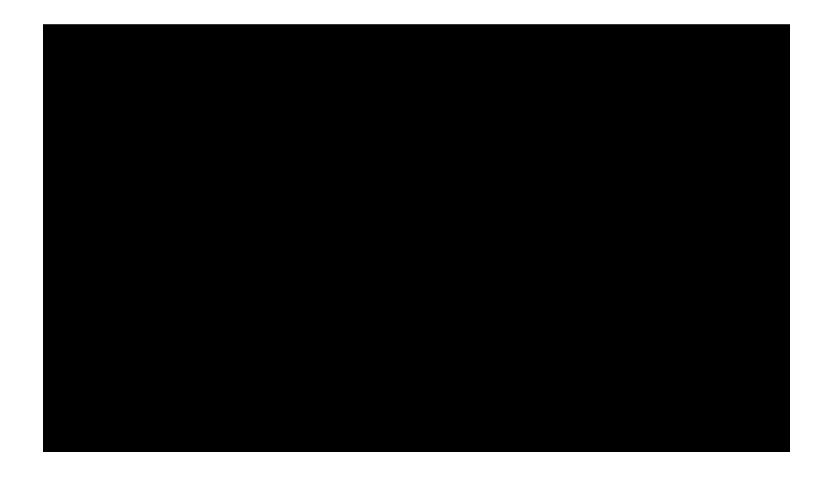
Prevention



MISTAKES

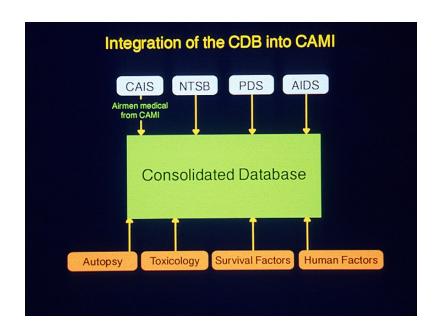
It Could Be that the Purpose of Your Life Is Only to Serve as a Warning to Others.

CAMI Safety Database



Aircraft Accident Safety Database

- FAS support
 - ground fatalities
- ATA
- Jessica Dubroff accident
- Cabin Safety, David Palmerton
- Autopilot related mishaps, Dr Beringer
- Toxicology support
- Update of the MI and Cardiovascular in flight mishaps



Methods

- Using the CAMI CDB
- Examine the NTSB database for spatia disorientation accidents (direct person codes) and for aircraft control not maintained



Methods

- NTSB data dictionary has listings of the coo used in the NTSE database
- SD code 33400 h note to also look 24566 Aircraft Control, 3127 not maintained

Methods (SQL)

WHERE NTSB_SOE_CAUSES.NSC_b_subject_code = 24566

AND ntsb_soe_causes.nsc_b_modifier_code = 3127

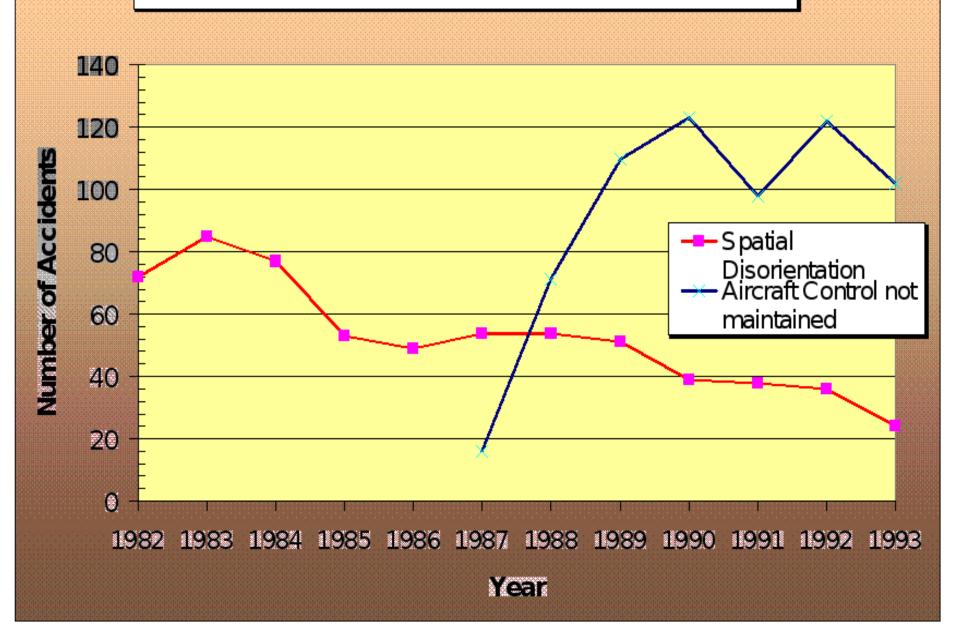
WHERE NTSB_SOE_CAUSES.NSC_DIRECT_CODE IN ('33400','43400','53400','63400')

Methods

- 1982 to 1993 available online, with data to 1999 to follow later
- SD cases = 632 over the time 1982-1993
- Aircraft Control/ not maintained cases = 642, however these codes only appear in 1987-1993 data

Year	Spatial Dis	Aircraft C		
1982	72			
1983	85			
1984	77			
1985	53			
1986	49			
1987	54			16
1988	54			71
1989	51			110
1990	39			123
1991	38			98
1992	36			122
1993	24			102
	Total		Total	
	632			642

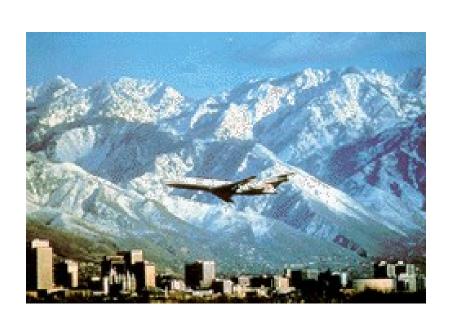




Results

- Many times the flight FAR field was empty
- Part 91
- Part 135 (1)
- Part 137 (2)
- Part 129 (2)
- Part 121 (3)





Part 121 cases

- DCA84AA002 HS-748-2A
- DCA84AA024
 Lockheed L-188
- DCA92MA022 DC-8-63

• DC-9 at Charlotte, NC July 1994

Part 121 cases

- DCA84AA002 HS-748-2A 10 fatal
- ABOUT 1.5 MIN AFTER DEPARTING SPRINGFIELD, IL, THE FLTCREW REPORTED A SLIGHT ELECTRICAL PROBLEM, BUT THEY CONTINUED ON COURSE. ABOUT 33 MIN LATER & A FEW MIN BEFORE THE ACFT SHOULD HAVE REACHED ITS DESTINATION, THE ACFT CRASHED. IMPACT OCCURRED WHILE THE ACFT WAS DESCENDING IN A RIGHT WING LOW ATTITUDE. BEFORE CRASHING, THE PLANE'S HEADING HAD CHANGED ABOUT 180 DEG. A CVR TRANSCRIPT REVEALED THE L GENERATOR (GEN) HAD FAILED AFTER TAKEOFF & THE 1ST OFFICER HAD MISTAKENLY ISOLATED THE R GEN. ATTEMPTS TO RESTORE THE R GEN WERE UNSUCCESSFUL. THE CAPTAIN ELECTED TO CONTINUE TO THE DESTINATION RATHER THAN RETURN TO THE NEARBY DEPT ARPT. THE CLD BASES WERE AT 2000' MSL, BUT ATC COULD NOT PROVIDE AN IFR CLNC BELOW 3000 FT. JUST BEFORE CRASHING, THE CREW INDICATED A TOTAL LOSS OF ELECTRICAL POWER. THE L GEN DRIVE SHAFT HAD SHEARED. THE REASON FOR THE R GEN NOT TO RESET WAS NOT DETERMINED. THERE WAS EVIDENCE THAT RECURRENT FLTCREW TRAINING DID NOT PREPARE THE CREW TO UNDERSTAND & COPE WITH THE ELEC PROBLEM & THAT FAA SURVEILLANCE DID NOT DETECT THE TRNG DEFICIENCY.
- Probable Cause

In-flight planning/decision..Improper..Pilot in command Spatial disorientation..Pilot in command

Contributing Factors

Electrical system, generator. Failure, partial
Electrical system, generator. Switched off
Electrical system, generator. Failure, total
Self-induced pressure. Pilot in command
Inadequate recurrent training. Company/operator management
Inadequate surveillance of operation. FAA(organization)

Part 121 cases

- DCA84AA024 Lockheed L-188 4 fatal
- AFTER DEPARTING THE BALTIMORE-WASHINGTON ARPT, THE ACFT HAD CLIMBED TO FL 220. ACCORDING TO INFO ON THE COCKPIT VOICE RECORDER (VCR), THE CREW EXPERIENCED GYRO PROBLEMS DURING THE CLIMB. THEY SELECTED THE #1 VERTICAL GYRO TO DRIVE BOTH APPROACH HORIZONS (ATTITUDE INDICATORS), SINCE THERE WAS AN INDICATION OF A MALFUNCTION IN THE #2 VERTICAL GYRO SYS. ABOUT 7 MIN AFTER LEVELING AT FL 220, THE FLT WAS CLEARED TO THE DRYER VOR. SHORTLY AFTER THAT, THERE WERE INDICATIONS OF CONFUSION IN THE COCKPIT WHICH INCLUDED THE STATEMENTS. "WHAT'S HAPPENING HERE," "YOU GOT IT?" & "NO." THE ACFT ENTERED A RIGHT DESCENDING SPIRAL; THE INDICATED AIRSPEED INCREASED FROM APRX 205 TO 317 KTS: THEN AN IN-FLT BREAKUP OCCURRED. THE WRECKAGE WAS FOUND SCATTERED OVER AN AREA APRX 2 MI LONG BY 1 MI WIDE. AN EXAM OF THE WRECKAGE REVEALED THAT THE IN-FLT STRUCTURAL FAILURE HAD OCCURRED DUE TO OVERLOAD, THEN A FIRE IGNITED IN THE RIGHT WING AFTER IT HAD FAILED. THE ACFT WAS NOT EQUIPPED WITH AN INDEPENDENT STANDBY ATTITUDE INDICATOR.
- Probable Cause

Flight/nav instruments, attitude gyro...Undetermined Airplane handling..Not maintained..Pilot in command Spatial disorientation..Pilot in command Design stress limits of aircraft..Exceeded..Pilot in command

Contributing Factors

Light condition..Dark night



- NYC91FA239 Convair 580
- ATL89MA072 HS-748-2A

1991 crash blamed on co-pilot

Report: Belvidere crash came in turn

By Shay Totten Free Press Staff Writer

co-pilot's mistake sent a Canadian cargo plane spiraling out of control 17 months ago, killing two people in a crash that scattered frozen fish and parcels across acres of farm

Canair cargo Flight 401 was carrying a load of frozen seafood and Federal Express packages to Hamilton, Ontario, from Moncton, New Brunswick, at 9:50 p.m. Sept. 18, 1991, when it broke apart in flight and crashed.

Pilot John McDougall, 30, of Mississauga, Ontario, and co-pilot Leonard Zilvytis, 31, of Mount Hope, Ontario, were killed, and their bodies were found the following morning.

According to a recently released National Transportation Safety Board investigation, Zilvytis was changing the plane's direction when he became disoriented during a 30-degree left turn. Instead of straightening out the plane, Zilvytis continued turning.

Investigators think he could not see the horizon and possibly became confused about direction because of an imbalance in his inner ear fluids, caused by the turn's force. He was turning and had nothing to anchor his gaze on, they said. At the same time, he was listening to a radio transmission of weather reports, which might have distracted him.

After several repetitions of the left turn, the plane entered into a "graveyard spiral," Stephen Veronneau, one of the investigators, said in the report. McDougall was checking on the cargo at the time of the crash, the report said.

The crash scattered wreckage over a 4-mile by ½-mile stretch between Cold Hollow and Laraway Mountains west of the North Branch River in Belvidere. The cockpit was found on Laraway Mountain along with the cockpit voice recorder. Frozen lobster was found as far as 6 miles away, the report said.

Gloria and Fred Allard, whose home is in the flight path, were among several residents who saw and reported the plane crash.

"The plane went right over our house," Gloria Allard said. "The tip part of the right wing was found on our property, and a tail section was found on the property above us."

"We were all sitting here watching television when we heard it going overhead. It was a tremendous roar and shook the house," Fred Allard said. "I ran outside to see what it was, and I saw it go across Route 109 and into the mountain."

Gordon Smith, fire chief of the Johnson Fire Department, said local rescuers didn't know what type of plane they were searching for or whether there were

Crash path of Canair Flight 401 Convair 580 Investigators of the National Transportation Safety Board have released an explanation of what caused Canair Flight 401 to break up in midflight over Belvidere on Sept. 18, 1991. Here's what they found: Direction of wind BELVIDERE 109 Tail cone Belvidere Right wing Lamoille Rive North Branch Left wing 9:48 p.m. Co-pilot flies Area of detail plane while pilot checks Doors E cargo. Co-pilot begins a 30-degree turn to the left to Rudder E 109 avoid traffic and a storm. The plane is cruising at about 260 mph debris found in red area Right propeller 53 seconds later MONTGOMERY Co-pilot tries to level the airplane but becomes 9:50 p.m. Co-pilot Left engine with propeller disoriented. continues turning the Cockpit plane left until it enters a 'graveyard Flight recorder spiral.' The aircraft reaches about 450 mph and begins to break apart. Radar contact is lost. Laraway Mountain 2,780 feet BELVIDER

Course NTCD

gers," Smith said. "You're trained for a fire, but you don't expect a commercial plane to come down in your area."

Rescue teams from Stowe, Cambridge and Johnson; the state police; Red Cross; Vermont Civil Air Patrol; and Lamoille Ambulance squad assisted in the search. Within two days, most of the plane was recovered, but small search teams combed the area for about a week, Smith said. Federal investigators and Canair officials arrived at the scene the CHRIS WILLIS, Free Press

Smith said. "They know that mountain like the back of their hands. So we had a firefighter with a radio in each group so we could keep track of everyone."

Eventually the searchers were sent home, left to wonder what caused the small plane crash.

A year and a half later, several said they were glad to find out what happened so they could put the nighttime disaster to rest.

"We'd always wondered what had

30

NYC91FA239 Convair 580

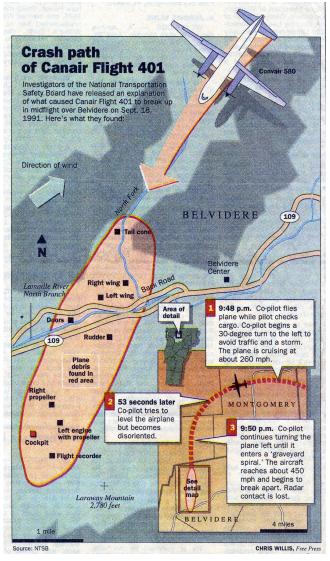




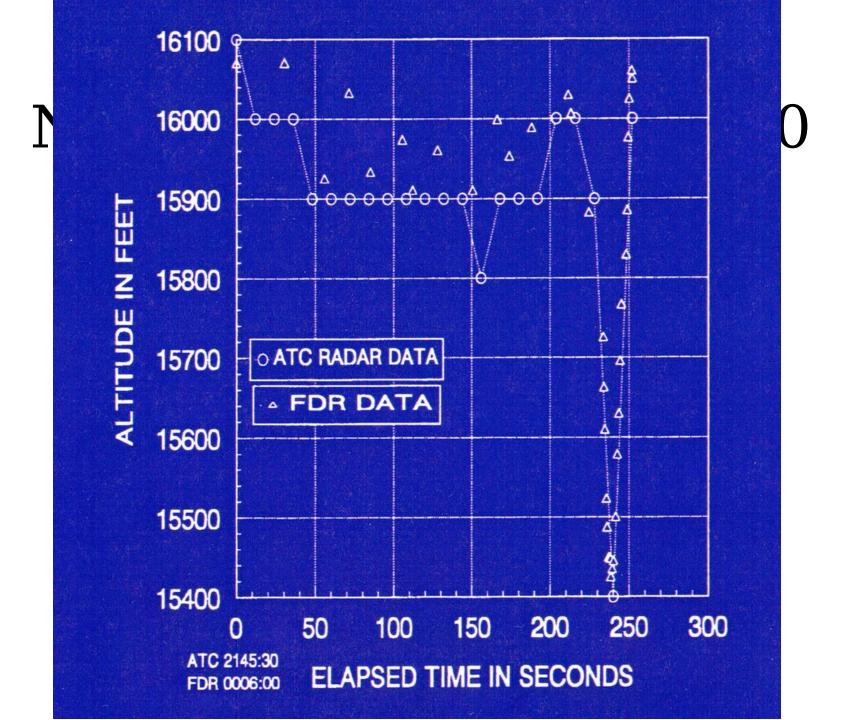
- 18-SEP-91 NYC91FA239
- THE AIRPLANE WAS CRUISING IN NIGHT INSTRUMENT METEOROLOGICAL CONDITIONS WHEN IT ENTERED A LEFT TURN AND EXCEEDED THE DESIGN AIRFRAME LIMITS. THE AIRPLANE BROKE UP IN THE DESCENT DUE TO AERODYNAMIC FORCES AND WAS DESTROYED. THE OUTBOARD WING PANELS HAD FAILED DOWNWARD AND CENTER WING SECTION SEPARATED FROM THE FUSELAGE. THE HORIZONTAL STABILIZER AND ELEVATORS HAD FAILED DOWN AND AFT. THE CAPTAIN WAS FOUND OUT OF THE COCKPIT WITH NO EVIDENCE OF HIM BEING IN THE SEAT AT IMPACT A HUMAN FACTORS STUDY FOUND THE AIRCRAFT'S LAST MINUTE OF FLIGHT MATCHED A PROFILE OF A PILOT EXPERIENCING SPATIAL

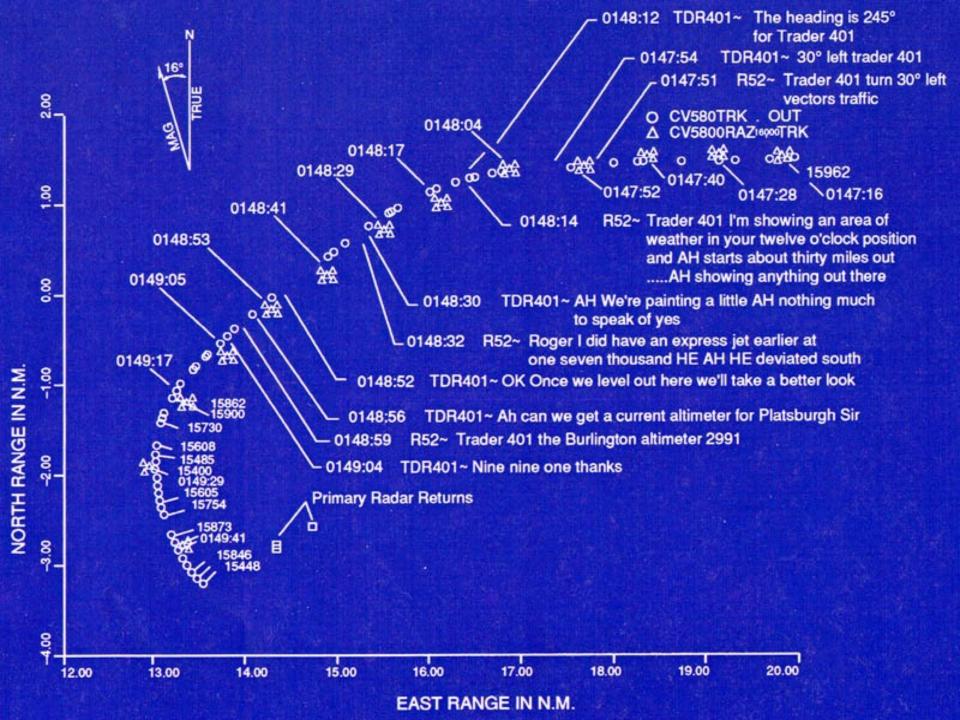


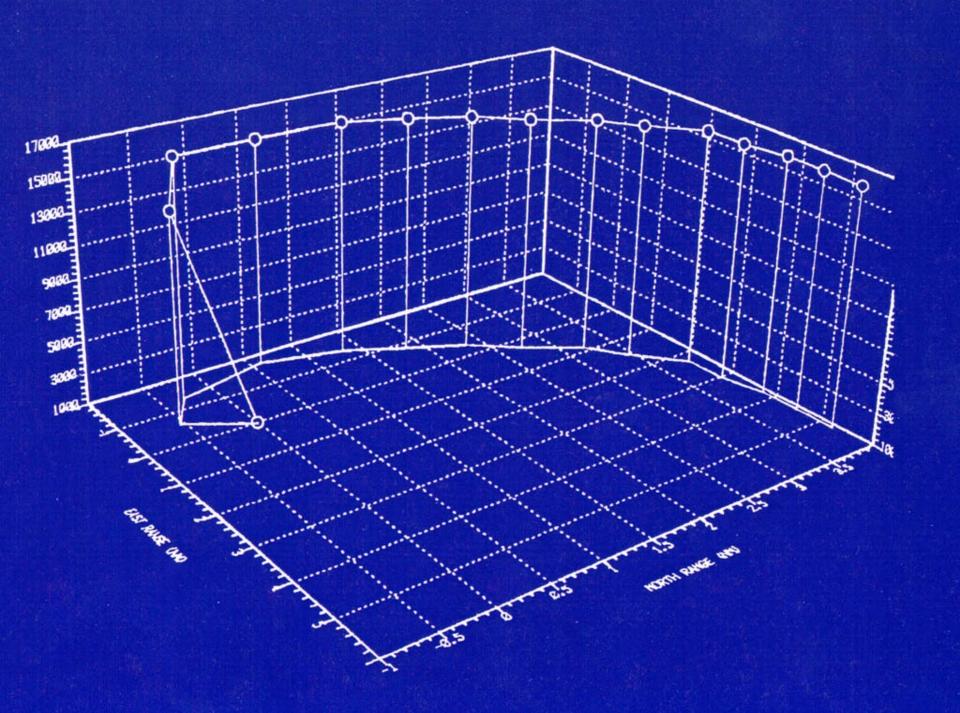
NYC91FA239 Convair 580



- 18-SEP-91 NYC91FA239
- Probable Cause FAILURE OF THE FIRST OFFICER (CO-PILOT) TO MAINTAIN CONTROL OF THE AIRCRAFT AFTER BECOMING SPATIALLY DISORIENTED, AND HIS EXCEEDING THE DESIGN STRESS LIMITS OF THE AIRCRAFT, FACTORS RELATED TO THE ACCIDENT WERE: THE LACK OF TWO PILOTS IN THE COCKPIT, DARKNESS, AND INSTRUMENT METEOROLOGICAL CONDITIONS (IMC) AT FLIGHT ALTITUDE.







This transcription is a rough draft. It is not the final certified copy.

Agencies Making Transmission

Abbreviation

Boston Center Montpieller Sector Radar Position

R52

Trader Four Zero One

TDR401

0147:51 R52

Trader four zero one turn ah thirty degrees left vectors traffic

0147:54 TDR401

Thirty degrees left Trader four zero one

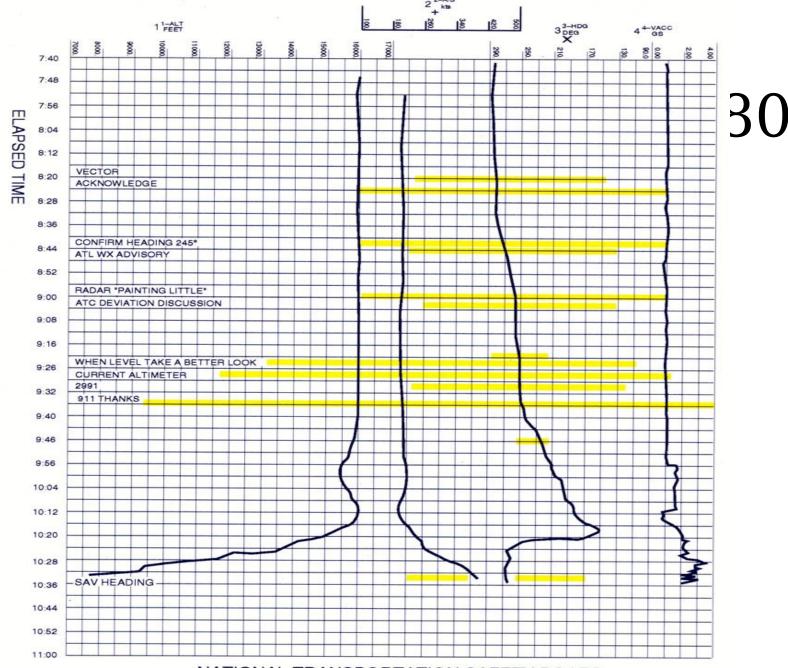
0148:12 TDR401

The heading is two forty five for Trader four zero one

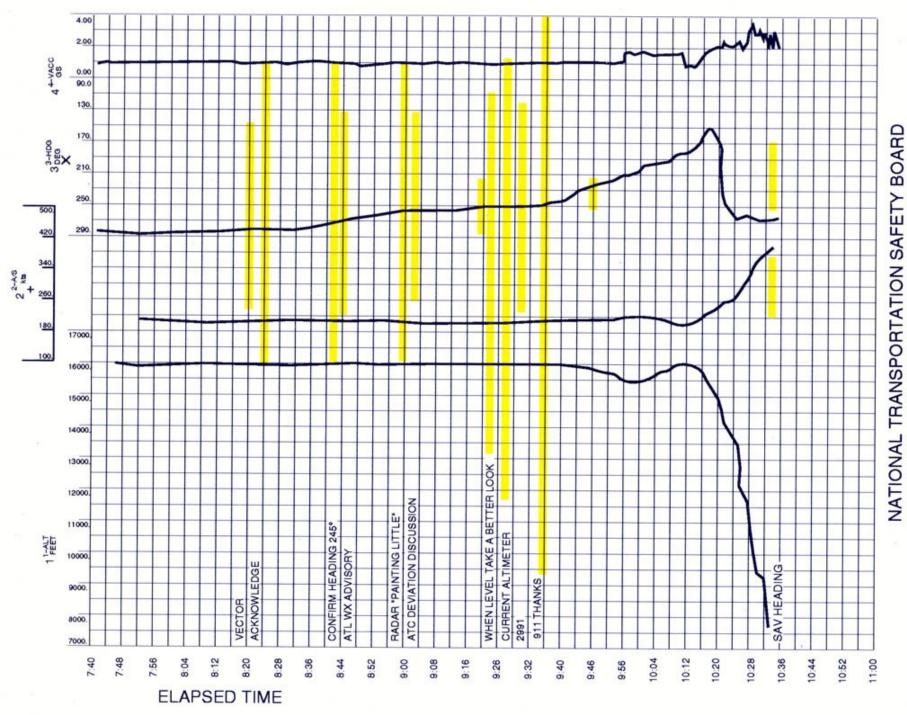
0148:14 R52

Trader four zero one roger I'm showing an area of weather in your twelve o'clock position and ah starts about thirty miles out extends ah a good seventy miles along your route of flight and also continues to your ten o'clock position ah between ten and twelve o'clock ah showing anything out there

0146:30 TDR401	Ah we're painting a little ah nothing much to speak of yes					
0148:32 R52	Roger I did have an express jet earlier at one seven thousand he ah he deviated southeast ah he went around the south side of it ah your route of flight takes you right through the center of it right now I'm not sure if you want to try going around the ah northwest side da the right or going around the south side to the left					
0148:52 TDR401	Okay once we level out here we'll take a better look					
0148:56 TDR401	Ah can we get a current altimeter for Platsburg air					
0148:59 R52	Trader four zero one the ah Burlington altimeter two niner nine one					
0149:04 TDR401	Nine nine one thanks					
0150:04 R52	Trader four zero one say heading					
0150:14 R52	Trader four zero one say heading					
0150:37 R52	Trader four zero one Boston Center					
0150:47 R52	Trader four zero one Boston how do you hear					
END OF TRANSCRIPT						



NATIONAL TRANSPORTATION SAFETY BOARD OFFICE OF RESEARCH AND ENGINEERING WASHINGTON, D. C.



OFFICE OF RESEARCH AND ENGINEERING WASHINGTON, D. C.

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26 48 24.00 15942. 219.2 239.15 -94.03 -0.24 1.04 0.00 -9.07 1.08 256.19 265.9 204.5 1.30 27 48 24.40 15943. 219.9 238.69 -46.66 -0.12 1.08 -0.01 -12.09 1.18 255.81 266.6 205.0 1.30 28 48 29.10 15954. 219.2 235.54 149.60 0.39 1.01 -0.01 -5.00 1.60 253.22 265.6 204.2 1.29 29 48 34.30 15967. 218.4 233.97 147.66 0.38 1.00 0.00 -2.60 1.60 251.93 264.6 203.4 1.29
26 48 24.00 15942. 219.2 239.15 -94.03 -0.24 1.04 0.00 -9.07 1.08 256.19 265.9 204.5 1.30 27 48 24.40 15943. 219.9 238.69 -46.66 -0.12 1.08 -0.01 -12.09 1.18 255.81 266.6 205.0 1.30 28 48 29.10 15954. 219.2 235.54 149.60 0.39 1.01 -0.01 -5.00 1.60 253.22 265.6 204.2 1.29 29 48 34.30 15967. 218.4 233.97 147.66 0.38 1.00 0.00 -2.60 1.60 251.93 264.6 203.4 1.29
27 48 24.40 15943. 219.9 238.69 -46.66 -0.12 1.08 -0.01 -12.09 1.18 255.81 266.6 205.0 1.30 28 48 29.10 15954. 219.2 235.54 149.60 0.39 1.01 -0.01 -5.00 1.60 253.22 265.6 204.2 1.29 29 48 34.30 15967. 218.4 233.97 147.66 0.38 1.00 0.00 -2.60 1.60 251.93 264.6 203.4 1.29
28 48 29.10 15954. 219.2 235.54 149.60 0.39 1.01 -0.01 -5.00 1.60 253.22 265.6 204.2 1.29 29 48 34.30 15967. 218.4 233.97 147.66 0.38 1.00 0.00 -2.60 1.60 251.93 264.6 203.4 1.29
29 10 31.30 13907. 218.1 233.97 147.66 0.38 1.00 0.00 -2.60 1.60 251.93 264.6 203.4 1.29
30 46 36.90 139/4. 21/.8 233.4/ 140.50 0.36 1.00 0.00 -2.14 1.59 251.52 264.0 202.9 1.29
22 40 50 00 15000 210 2 222 22 222 222 222 222 222 222
77 40 EE 70 16007 100 7 100 CF 44 14
34 48 59.50 16014. 221.2 227.98 -36.66 -0.09 0.95 0.00 -6.07 1.18 246.93 266.6 204.4 1.27
35 49 1.50 16018. 221.4 226.95 -278.68 -0.71 0.97 -0.01 -4.97 0.68 246.07 266.6 201.4
36 49 3.30 15995. 222.0 226.30 -585.57 -1.49 0.97 0.03 -5.36 0.03 245.50 267.0 205.2 1.27
734 40 6 00 15067 334 1 334 67 464 74
30 33 0.30 13307. 224.1 224.07 -404.78 -1.17 0.99 0.03 -10.36 0.28 244.14 268.7 206.6 1.28
1.05 0.06 0.02 -13.00 -0.31 241.83 268.2 206.3 1.27
41 49 13.10 15893. 230.8 218.04 -900.93 -2.20 1.09 0.02 -16.02 -0.62 238.48 273.4 210.5 1.20
42 49 14.60 15871. 231.4 215.56 -695.05 -1.70 1.21 0.07 -26.51 -0.25 236.39 273.1 210.4 1.32
43 49 15.20 15865. 231.9 213.97 -597.56 -1.46 1.37 0.05 -40.02 -0.19 235.05 272.9 210.3 1.37
13 13 13 13 13 13 13 13 13 13 13 13 13 1
13 19 16.20 13636. 233.8 208.45 -731.02 -1.77 1.37 -0.05 -52.61 -0.68 230.31 272.8 210.2 1.37
47 40 to 00 temps 224 4 201 th 244 th
49 49 20.30 15730. 241.6 196.25 -1980.93 -4.62 1.32 -0.08 -40.62 -3.05 219.54 274.9 212.3 1.35
50 49 20.50 15721. 244.4 195.77 -1677.27 -3.87 1.41 -0.07 -31.97 -2.26 219.05 277.2 214.2 1.36
51 49 23.80 15714. 244.4 187.65 -7471.67 -16.78 1.53 -0.17 -30.86 -13.97 211.60 281.8 217.8 1.38
52 49 24.10 15651. 245.7 186.71 -7294.61 -16.32 1.47 -0.17 -24.38 -13.55 210.72 281.8 218.0 1.37
33 43 21.30 13000. 243.2 103.14 -0041./2 -14./3 2.00 0.18 -27.11 -12.08 209.22 282.5 218.7 1.49
55 49 26.00 15485. 255.1 181.76 -4485.38 -9.84 2.18 0.00 -2.61 -7.50 206.07 281.6 218.5 1.54 56 49 26.10 15476. 259.4 181.58 -4198.52 -9.07 2.22 -0.04 -6.08 -6.82 205.80 285.4 221.5 1.54
57 49 26.80 15452. 260.7 182.90 -1925.14 -4.17 / 2.32 0.07 1.46 / -2.26 207.08 205.0 231.3 1.66
58 49 27.20 15438. 261.3 182.62 -1210.75 -2.62 1.85 0.09 -9.00 -0.97 206.83 285.0 221.4 1.45

1).

PRINTOUT OF OUTPUT DATA

POINT	MIN	SEC	ALTITUDE FT	GROUND SPEED KNOTS	TRACK ANGLE DEG	VERT. VEL. FPM	FLIGHT PATH DEG	LIFT G,S	T-D GS	ROLL DEG	PITCH DEG	HEADING DEG MAG	AIRSP: TRUE KNOTS	IND. KNOTS	AOA
59 60 61 62 63 64 65 66 67 68 69 70	49 49 49 49 49 49 49 49	27.60 28.70 29.00 29.40 30.20 30.60 31.30 31.50 32.10 32.50 33.10 33.30 33.90	15437. 15415. 15418. 15423. 15433. 15450. 15480. 15536. 15567. 15605. 15618. 15657.	265.7 266.5 267.2 267.4 268.4 268.8 269.2 267.0 265.5 261.0 260.8 260.7	181.78 176.97 175.72 174.02 170.33 168.72 166.92 166.62 166.62 164.73 164.80	-787.39 52.61 335.18 789.05 1834.57 2277.10 3059.77 3436.83 4507.82 4458.59 3852.30 4054.55 3651.11	-1.67 0.11 0.71 1.67 3.86 4.78 6.40 7.24 9.51 9.56 8.29 8.72 7.86	1.74 1.80 1.84 1.97 1.76 1.82 1.88 1.64 1.49 0.97 0.97 0.99	0.08 -0.05 -0.07 0.02 0.04 -0.18 -0.05 -0.13 -0.13 -0.11 0.07	-21.55 -34.01 -35.45 -33.09 -32.35 -25.21 -7.99 -9.68 -7.89 -18.77 -15.26 -15.84 -24.37	-0.23 1.29 1.84 2.80 4.86 5.85 7.56 8.31 10.45 10.33 9.17 9.57 8.68	205.98 201.59 200.43 198.85 195.46 193.96 192.30 192.41 192.22 191.84 190.58 190.65 189.63	288.7 286.0 285.8 285.0 283.7 283.3 283.1 281.4 281.3 276.5 274.8 273.5	224.3 222.2 222.1 221.4 220.3 219.9 219.7 218.3 218.1 214.1 212.3 212.6 211.4	1.41 1.43 1.44 1.48 1.44 1.46 1.41 1.37 1.25 1.26
72 73 74 75 76	49 49 49 49	35.20 38.10 38.30 38.90 39.70	15754. 15818. 15834. 15889. 15985.	255.9 256.0 254.9 251.3 252.6	161.95 158.07 158.08 155.44 153.43	2977.17 4133.35 4551.73 6174.33 5483.10	6.55 9.05 9.99 13.62 12.08	0.93 2.11 1.62 1.93 0.45	-0.12 0.09 0.05 0.21 0.04	-18.41 -12.05 -23.24 -25.68 -91.83	7.51 10.40 11.09 14.80 11.91	188.12 184.55 184.61 182.35 180.24	266.8 265.5 265.1 262.9 261.3	205.8 204.6 204.3 202.3 200.8	1.26 1.60 1.46 1.56 1.13
777 78 79 80 81 82 83 * 84 * 85 * 86	49 49 49 49 49	40.70 41.20 42.00 42.80 43.90 44.10 44.80 45.60 46.10 47.10	16049. 16045. 16038. 15998. 15925. 15910. 15846. 15762. 15653. 15448.	253.7 255.8 257.5 262.0 261.9 265.7 273.7 273.3 273.3	151.89 147.83 146.59 142.48 135.92 132.91-	1776.78 -88.69 -2085.57 -3260.30 -4592.57 -4521.38 -6086.72 -9975.08 10906.62 11632.71	3.95 -0.20 -4.57 -7.00 -9.81 -9.53 -12.37 -19.80 -21.49 -22.68	0.55 0.53 0.20 0.71 1.32 1.35 1.88 1.66 1.65	0.11 - 0.14 - 0.27 0.11 0.19 0.11 -		2.80 -1.33 -4.87 -6.42 -9.52 -9.59 -13.17 -21.01 -23.02 -22.72	179.14 179.02 179.02 178.00 173.89 172.50 167.96 161.05 157.83 156.32	256.6 258.1 260.7 265.7 264.0 266.6 273.9 279.3 280.4 283.2	196.9 198.0 200.2 204.1 203.1 205.2 211.1 215.7 216.9 219.8	1.17 1.16 1.06 1.20 1.38 1.38 1.50 1.42 1.42

^{*} SMOOTHED VALUES ARE APPROXIMATE NEAR END POINTS

Part 129

- ATL89MA072 HS-748-2A 2 fatal
- DRG NGT CARGO OPN, CHECK CAPT (RGT SEAT) WAS EVALUATING THE 1ST OFFICER (F/O. LEFT SEAT) FOR PSBL UPGRADE TO CAPT. BFR DEPG. FLT WAS CLRD FOR RGT TURN AFTER TKOF TO 020 DEG. TKOF BGN AT 0441:11. WTR/METHANOL INJECTION WAS USED (TO 1ST PWR RDCN). AT 0441:49, LNDG GEAR WAS RETRACTED; 8 SEC LTR 1ST PWR RDCN WAS MADE, THEN A FREQ CHG WAS APPROVED. CAPT NOTED THEY SHLD CLB TO 1500' MSL (APRX 500' AGL) BFR TURNING. AT ABT 300' AGL, ACFT ENTERED OVC & BGN A STEEP RGT TURN. CVR INDCD CAPT WAS PERFORMING COCKPIT DUTIES AT THIS TIME & GIVING INFO TO F/O ABT THE DEP. FDR SHOWED ACFT RCHD MAX ALT OF 423' AGL & BGN DSCNDG. AT 0442:22. CAPT REMARKED TO F/O, "DON'T GO DOWN . . . GET UP . . . UP UP UP . . . UP, OH!" AT ABT THAT TIME, ACFT HIT IN AN OPEN FLD, BUT CONTD FLYING FOR APRX 3/4 MI. IT THEN HIT A TREE & CRASHED IN A WOODED AREA. INV REVEALED THAT DRG SVRL TRNG FLTS & 2 CHECK FLTS, THE F/O DEMONSTRATED DIFFICULTY IN PERFORMING INSTRUMENT FLT DUE TO DISORIENTATION, NARROW FOCUS OF ATTENTION, OR LACK OF INSTRUMENT SCAN (INST FIXATION), ESPECIALLY DRG HI TASK WORK LOAD.

Probable Cause

IMPROPER IFR PROCEDURE BY THE FIRST OFFICER (COPILOT) DURING TAKEOFF, HIS LACK OF INSTRUMENT SCAN (IMPROPER USE OF FLIGHT/NAVIGATION INSTRUMENTS). HIS FAILURE TO MAINTAIN A POSITIVE RATE OF CLIMB OR TO IDENTIFY THE RESULTANT DESCENT, AND THE CAPTAIN'S INADEQUATE SUPERVISION OF THE FLIGHT. CONTRIBUTING FACTORS WERE: DARK NIGHT, LOW CEILING, DRIZZLE, THE FIRST OFFICER'S LACK OF TOTAL EXPERIENCE IN THE TYPE

OF OPERATION, AND POSSIBLE SPATIAL DISORIENTATION OF THE FIRST OFFICER.

• Part 135

- CHI84FA058 C207 4 fatal
- THE PLT & 3 PASSENGERS TOOK OFF AT NIGHT ON AN OVER WATER FLT TO AN ISLAND IN LAKE ERIE TO PROVIDE HELP TO A HEART PATIENT. NO FLT PLAN WAS FILED & NO RECORD OF A WX BRIEFING WAS FOUND. REPORTEDLY, AFTER TAKEOFF, THE ACFT DISAPPEARED IN A CLOUD OR HAZE. ALSO AFTER DEPARTING, SHERIFF'S PERSONNEL RECEIVED A RADIO CALL FROM THE ACFT STATING "WE ARE IN IT." ACCORDING TO LOCAL RESIDENTS, THERE WAS PATCHY FOG IN THE AREA. WHEN THE ACFT DID NOT ARRIVE AT ITS DESTINATION, A SEARCH WAS INITIATED. THE PLANE WAS FOUND IN LAKE ERIE ALONG THE EXPECTED ROUTE OF FLT. THE ACFT WAS INTACT, EXCEPT THE ENG WAS LOOSE FROM THE FIREWALL & THERE WAS MAJOR DAMAGE TO THE RIGHT, OUTER WING PANEL. NO PREIMPACT/MECHANICAL MALFUNCTION/FAILURE WAS FOUND. ABOUT 45 MI WEST AT TOLEDO, OH, THE 2150 WX IN PART WAS: 1500 FT OVERCAST, VISIBILITY VARIABLE 1 TO 2 MI WITH FOG, TEMP 32, DEW POINT 31, WIND FROM 330 DEG AT 4 KTS.
- Probable Cause

Preflight planning/preparation..Inadequate..Pilot in command VFR flight into IMC..Continued..Pilot in command Proper altitude..Not maintained..Pilot in command Clearance..Misjudged..Pilot in command

Contributing Factors

Light condition..Dark night Weather condition..Fog Self-induced pressure..Pilot in command

Spatial disorientation..Pilot in command





- LAX88DUJ05 G164B
- DEN92LA062 AT-



Part 137

- LAX88DUJ05 G164B 1 minor
- DURING AN AERIAL APPLICATION FLIGHT, THE PILOT DISPENSED THE LOAD AND WAS RETURNING TO THE AIRSTRIP DUE TO AN APPROACHING THUNDERSTORM.
 ENROUTE, THE VISIBILITY WAS REDUCED TO NEAR ZERO FROM BLOWING DUST. THE PILOT LOST CONTROL OF THE AIRPLANE AND COLLIDED WITH THE TERRAIN. THERE WERE NO REPORTED MECHANICAL FAILURES OR MALFUNCTIONS AT THE TIME OF THE ACCIDENT.
- Probable Cause

In-flight planning/decision..Poor..Pilot in command Flight into known adverse weather..Intentional..Pilot in command

Contributing Factors

Weather condition..Sand/dust storm Spatial disorientation..Pilot in command

- DEN92LA062 AT-301 1 fatal
- WHILE FLYING BETWEEN FIELDS ON AN AERIAL APPLICATION FLIGHT, THE AIRCRAFT IMPACTED THE GROUND IN A SHALLOW DIVE WITH POWER ON THE ENGINE. WEATHER AT THE TIME WAS 800 FEET OVERCAST SKIES AND GROUND FOG. Probable Cause THE PILOTS FAILURE TO MAINTAIN ALTITUDE DUE TO SPATIAL DISORIENTATION. FACTORS WERE: LOW OVERCAST SKIES AND GROUND FOG.

- 1999 Nall Report http://www.aopa.org/asf/publications/99nall.html
- In 1998, six accidents contained specific references to spatial disorientation in the sequence of events or narrative sections of their reports. This number is, however, what statisticians call a "lower bound" on the true number of accidents in which spatial disorientation was a significant factor. The conditions surrounding a number of other weather-related accidents suggest that spatial disorientation might have been contributory there as well.

- 1999 Nall Report http://www.aopa.org/asf/publications/99nall.html
- A detailed analysis of accidents over a tenyear period (1987-1996) with an emphasis on spatial disorientation as a cause or significant contributory factor reveals a much higher involvement of this factor than suggested by the direct references in the 1998 reports. During this period, there was an average of almost 37.6 accidents per year, of which 33.9 were fatal. At this rate, there is one fatal spatial disorientation accident every eleven days. Over 90 percent of all the accidents during this time in which spatial disorientation was a factor resulted in fatalities.

Part 91

- 1999 Nall Report http://www.aopa.org/asf/publications/99nall.html
- Typically, these accidents are suffered by noninstrument-rated pilots attempting to complete VFR flights in instrument meteorological conditions. At least one accident in 1998, however, occurred when an experienced instrument-rated pilot in a wellequipped turbine-powered airplane became disoriented during the visual portion of a In this case <u>annroach</u>

circling IFR night exace

 Conditions
 Total
 Fatal
 Percent Fatal

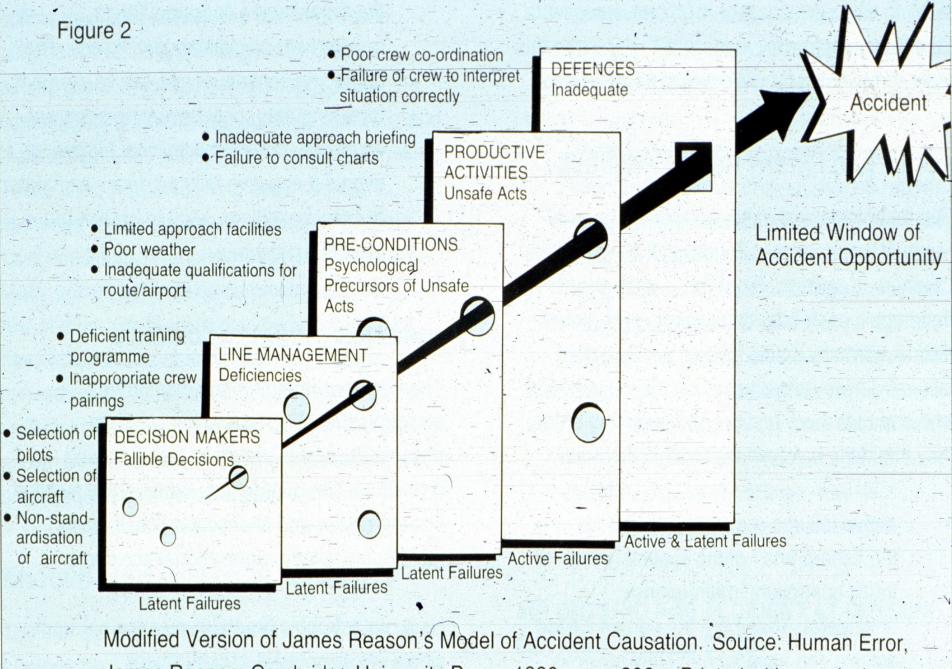
 All
 1,679
 341
 20.3%

 Day VMC
 1,216
 139
 11.4%

 Night VMC
 75
 18
 24.0%

 Day IMC
 58
 37
 63.8%

 Night IMC
 19
 13
 68.4%

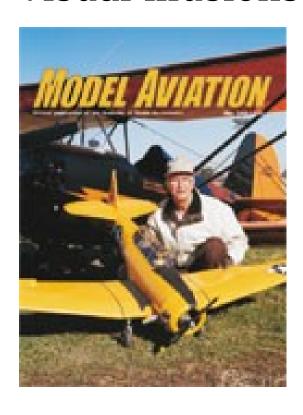


James Reason, Cambridge University Press, 1990, page 302. Printed with permission.



Human Factors

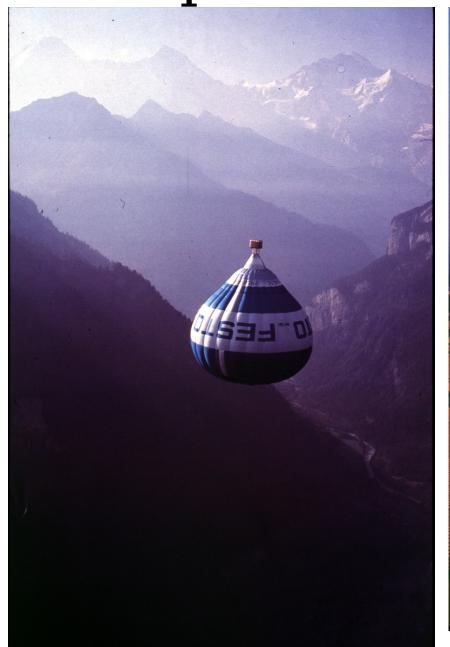
- Radio controlled aircraft runway
- Visual illusions







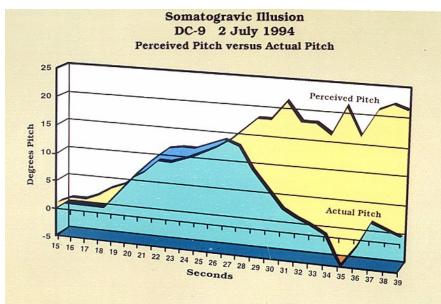
Spatial Orientation

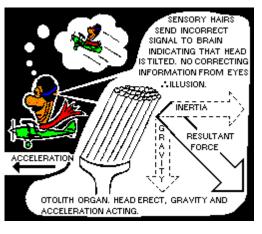




- DC-9 missed approach
- Windshear
- ATC info provided





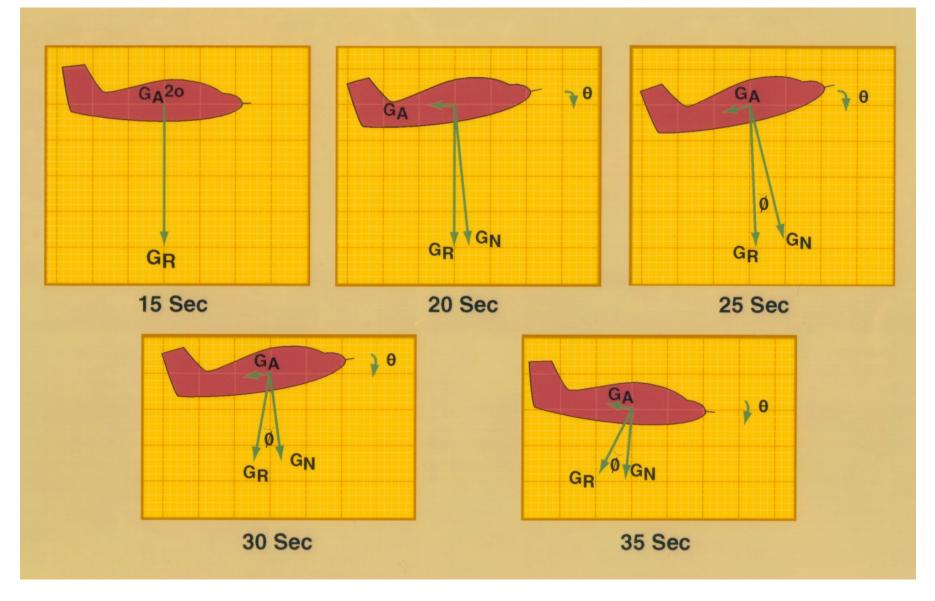


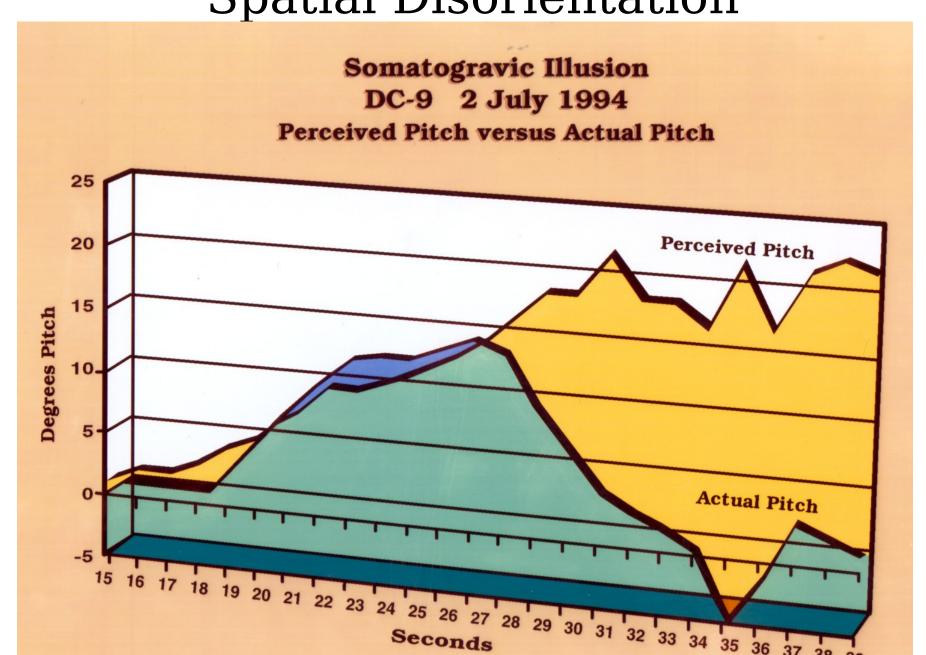
- DC-9 missed approach
- Windshear
- ATC info provided



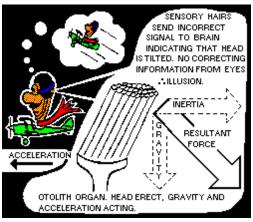


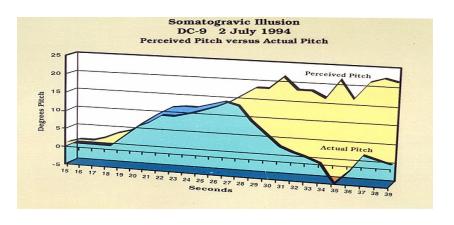




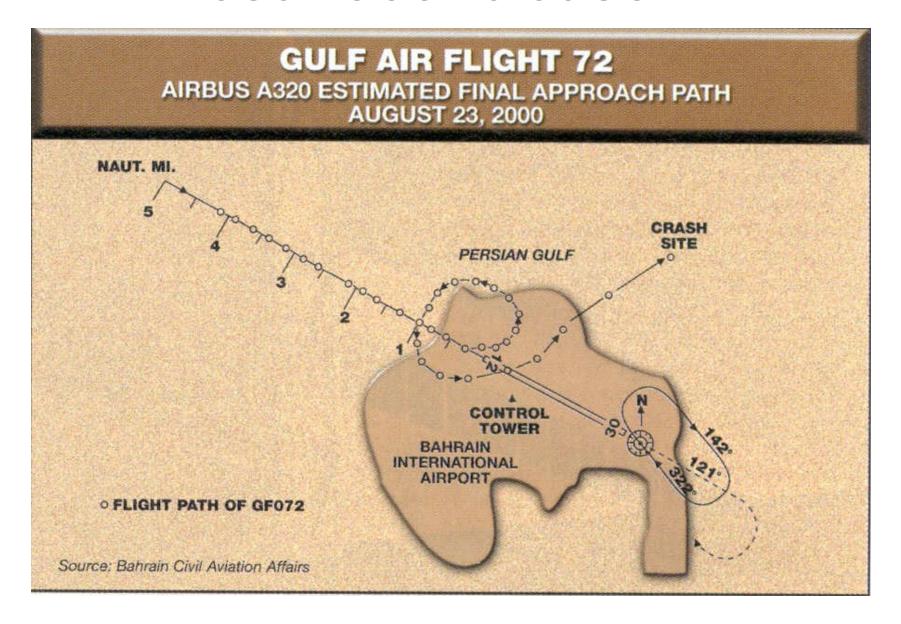








Most recent case



- Lake Victoria
- Black hole approach
- Narrow runway
- ATC warned





Emergency

• Smoke in the cockpit



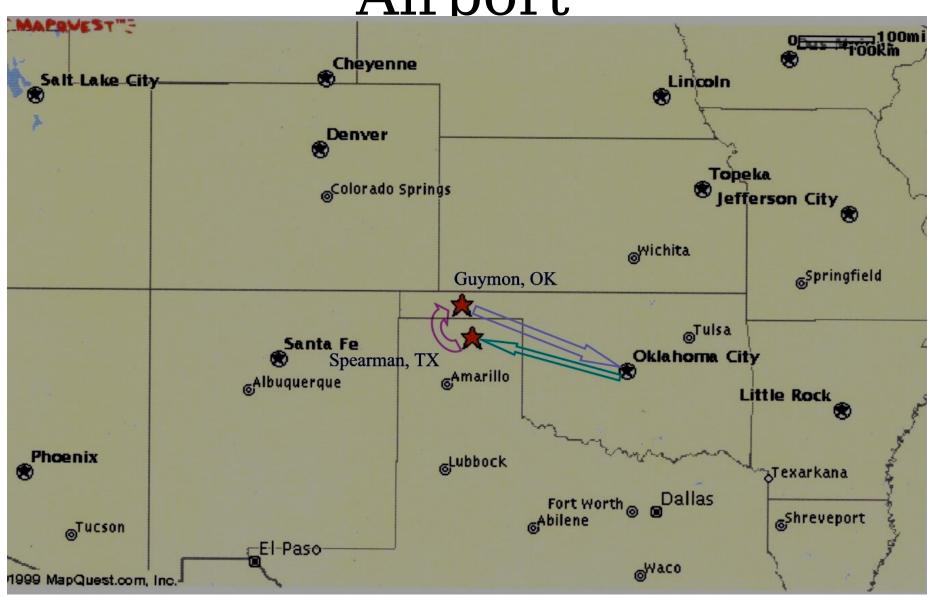


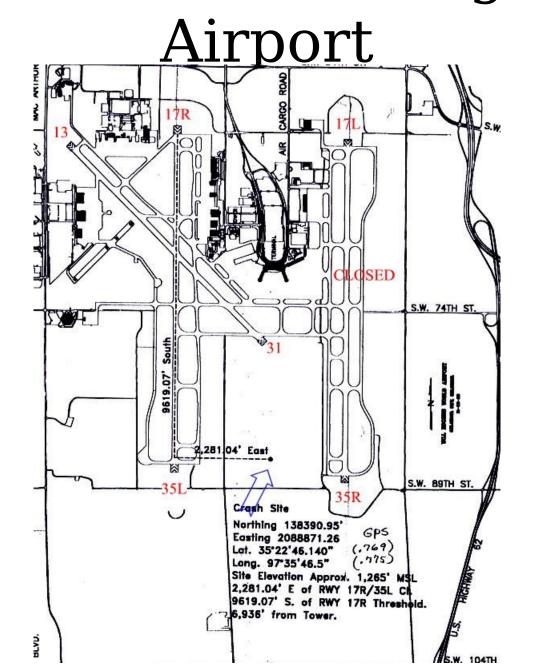
C210 N94 Will Rogers Airport 25 October 2000

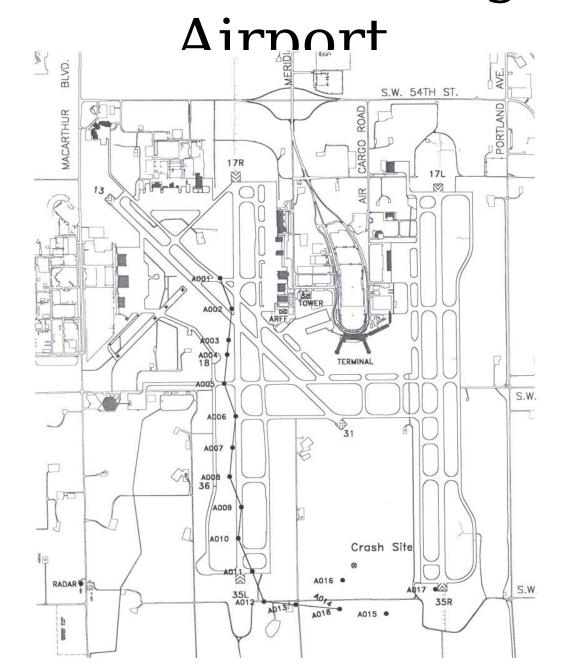


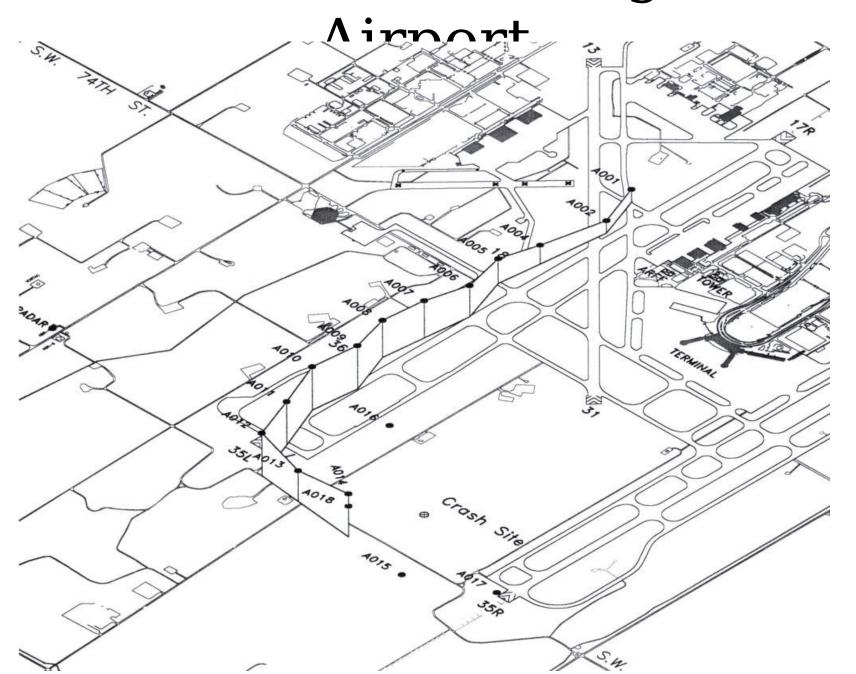


C210 N94 Will Rogers Airport









C210 N94 Will Rogers Airport

INDEX	LATITUDE	LONGITUDE
"A1" "A2" "A4" "A5" "A6" "A7" "A8" "A9" "A10" "A11" "A12" "A13"	35.235765089 35.235009041 35.233874098 35.233157287 35.232341505 35.231564960 35.230848156 35.230092097 35.225315563 35.224493228 35.223737174 35.223652111	97.362563249 97.362221383 97.362373156 97.362469008 97.362135174 97.362239014 97.362239014 97.362334862 97.361993060 97.361993060 97.361690170 97.361348414 97.360400285
"A14" "A16"	35.223534298 35.224251095	97.355087490 97.354991575
"A18"	35.223534298	97.355087490















Airnort

RACKING DATA TIME SCAN	ACID	TRK	ABC		0/26. FRM	RALT	PACP		PAGE 2 PRAN	x .	Υ.	DDEG	DRAN	χv	YV	HDG	SPD HI	DI	ADS	С	SYS
00:43:04.814	M9485T	76	6235		7		0	. 0	0.00	7.94	-7.94	135	11.22	9	0	90	9		DEP	14	1
00:43:05.500	N9485T	83	6235	6235	33	1500	233	20	1.43	0.44	1.31	18	1.38	11	-43	165	44		DEP	14	
171 00:43:10.125	N9485T	83	6235	6235	34	1500	253	22	1.31	0.50	1.19	55	1.29	3	-53	176	53		DEP	18	0
172	N9485T	83	6235	6235	35		292	26	1.21	0.50	1.06	25	1.17	9	-59	171	59		DEP	16	0
173	N9485T	83	6235	6235	36	1600	332	29	1.10	0.50	1.00	26	1.12	8	-65	173	65		DEP	10	
174	N9485T	83	6235	6235	37	1700	386	34	1.00	0.50	0.88	29	1.00	10		171	70		DEP	653.49	
175	N9485T	83	6235	6235	38	1800	449	39	0.90	0.56	0.75	36	0.94	9	3302	173	1/2/200		DEP	117111	0
176	N9485T	83		6235		1900	531	47	0.82	0.56	0.62	42	0.84	12		171	2000	100	DEP	(5)(5)	0
177	N9485T	83	100000000	6235	1000	1900	628	55	0.75	0.56	0.50	48	0.75	11		172			DEP		0
178	N9485T	83		6235		2000	747	66	0.71	0.62	0.38	59	0.73	16		169	83			ACTION	
179	N9485T	19		6235		2000	877	77	0.69	0.62	0.25	68	0.67				2000	20	DEP		٥
180.	N948ST		CONTRACTOR	6235	2000	2000	Constants.		0.71			79		15		170	86.		DÉb	•	•
181	PERCENTAGE.	dr. ct	•			70.50	1012	89		.0.69	21.0	25,453	0.70	20	10.040.00	167	89		DEP		0.
182	N9485T	92		6235		2000	1108	97	0.83	0.75	0.00	. 90	0.75	42		153	73	1165	DEP	07470	0
183	N9485T	83	35300.0	6235		1800	1123	.99	1.03	0.88	0.00	90	0.88	78	-64	129	100		DEP	110	0
184	N7485T	83	1010-12-12-11	4235	12202	2000	1040	91	1.27	1.06	0.00	90	1.06	124	-1	90	124	1	DEP	1W	0
00:44:11.313 185	N9485T	83	6235		35	CST	1039	91	1.45	1.25	0.00	90	1.25	124	-1	90	124	ı	DEP	14	. 0
00:44:15.873	N9485T	43	6235	6235	36	2800	757	84	1.49	1.06	0.12	83	1.07	112	33	73	116		DEP	14	0
00:44:20.312 187	N9485T	23	4235		33	CST	946	83	1.66	1.44	0.12	85	1.44	112	33	73	116	1	DEP	14	0
00:44:24.937	N9485T	83	4235	4235	23	1800	854	75	0.98	1.06	0.12	83	1.07	-49	37	307	61	ľ	DEP	14	0
00144129.499	N9485T	83	6235		20	CST	806	71	0.93	0.94	0.25	75	0.97	-49	37	307	61	1	DEP	1 W	0
0144:34.124	N9485T	83	6235		18	CST	749	66	0.88	0.88	0.25	74	0.91	-49	37	307	61	1	DEP	14	0
0144138.564	N9485T	83	6235		16	CST	687	60	0.84	0.75	0.31	67	0.81	-49	37	307	61	I	DEP	14	0
0:44:43.189	N9485T	83	6235		15	CST	620	54	0.81	0.69	0.38	61	0.78	-49	37	307	61		DEP	14	. 0
0144147.488	N9485T	83	6235		14	CST	549	48	0.79	0.62	0.44	55	0.76	-49	37	307	61	I	DEP	14	0
0144152.313	N7485T	83	6235		13	CST	473	42	0.78	0.56	0.50	48	0.75	-49	37	307	61	D	EP	14	ο.
0144:56.815	N7485T	83	6235		12	CST	397	35	0.78	0.50	0.56	41	0.75	-49	37	307	61	D	EP	1W	0
0:45:01.438	N9485T	83	6235		12	CST	355	85	0.79	0.44	0.62	35	0.76	-49	37	307	61		EP	14	0
94	N9485T	83	4235		12	CST	248	22	0.80	0.31	0.69	24	0.75	-49		307	61	500	EP		

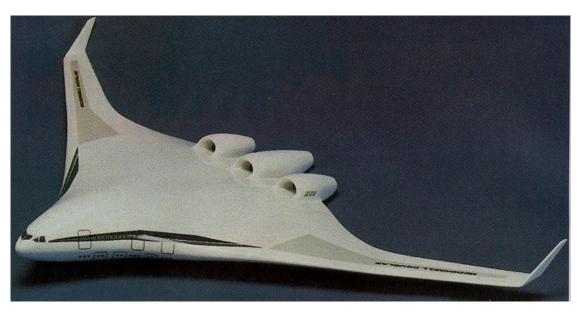
CFIT Prevention





What's Next?











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